

Design and Development of Material Handling Crane

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Abstract: - In case of construction equipment there is a mainly requirement of material to be transfer from one place to another place or from one floor to another floor. This material may contain the grain particles, cubes, heavy cement bags, etc. This is very important to transfer the material at right place at right time for its good working environment.

So there is a main requirement of the construction crane for the purpose of material handling from one place to another place.

This project will helpful for the specific requirement of such application of material handling in case of construction sites. Basically these construction sites requires more human interfere for proper completion of the construction work.

Keywords: Tower cum Jib crane, Steel structural design, FEM, Material sourcing, manufacturing procedure, testing procedure.

INTRODUCTION:

In the construction or heavy mechanical industry there is a need to transmit the material from one place to another place. There are so many varieties of cranes used in these industries. The commonly used cranes are overhead cranes, mobile cranes, rough terrain cranes, tower cranes, jib cranes. In case of construction work small cranes are normally used which carries approximately 100 kg can load. Also in this construction industry, the tower cranes are mostly used for high buildings applications. So there is a need of design of a crane, which considers the requirement of sponsorer.

Due to some limitations of conventional methods, it is impossible to improve workability. In order to successfully come up with these problems, material handling crane which is designed features plays vital role in day-today industrial as well as civil work. To overcome the problem in existing methodology of working it was proposed to **Design and development a Material handling Crane** by which the workability improves.

In this paper the Design, Manufacturing and Execution of crane. And proposed requirements of the project,

- Material of weight 500 Kg to be carry.
- Maximum lifting height should be 132 feet.
- Material can place in
- Motor drive for hoist to lift the material.
- Minimum persons required to operate the crane.

- Crane should be mobile having less weight.
- Easily assembled on construction site (normally within 2.5 hours).
- Angular movement of 180 degrees of crane axis.
- between radius of 20feet

ORGANIZATION OF THE WORK

Chapter 1: Design

This chapter gives various Design considerations in structures, components and its applications of CAD / CAE in new product design.

1.1 Calculations of the maximum stresses induced in the members of the vertical structure of the crane.

A structural failure may be said to occur when a structure collapses due to rupture or excessive distortion of members. The main mode for structural failure of material would be by shear, tension, buckling, crushing, fatigue or brittle fracture. In elastic design, the allowable stresses are taken less than the structure of material by an appropriate factor called "Factor of Safety".

- a) For steel, a factor of safety of **1.67** is used in axial tension and compression and **1.5** in bending.
- b) The usual value of load factor is taken **1.7** for beams.

The shear force at distance 6.3 meters from the free end is equal to the unbalanced vertical force i.e. $F_x = -10\text{KN}$ and the bending moment at this section,
 $M = -10\text{ KN} \times 6.3\text{ m} = -63\text{ KN-m}$

Box type design is considered, 500 mm x 500 mm is used as a column, with a one end fixed and other is free, then by Euler's formula the crippling load will be,

$$P = \frac{\pi^2 \times E \times I}{L^2} \quad [1]$$

Considering Modulus of Elasticity $E = 200\text{ GPa}$

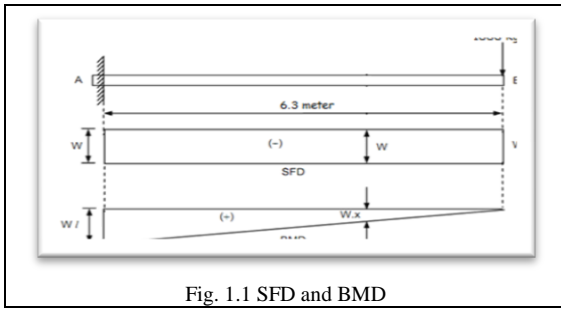


Fig. 1.1 SFD and BMD

So, calculating the Moment of Inertia for the selected column section will be,

$$I = \frac{BD^3}{12} = \frac{500 \times 500^3}{12} = 5.2 \times 10^9 \text{ mm}^4$$

$$\text{Safe Load} = \frac{\text{Buckling Load}}{F. O. S.} = 2.78 \times 10^6 \text{ KN}$$

Since, the column is fixed at one end and free at other end, the equivalent length of column is,

$$L = 2 \times l \text{ (length)} = 2 \times 3 \times 10^3 \text{ mm} = 6 \times 10^3 \text{ mm}$$

Sr. No	End Connections	Crippling Load	Relation between equivalent and actual length
1	Both ends hinged	$\frac{\pi^2 EI}{L^2}$	$L=1$
2	One end fixed and the other free	$\frac{\pi^2 EI}{2L^2}$	$L=2l$
3	Both end fixed	$\frac{\pi^2 EI}{(L/2)^2}$	$L=\frac{l}{2}$
4	One end fixed and other hinged	$\frac{\pi^2 \times E \times I}{(L/\sqrt{2})^2}$	$L=\frac{1}{\sqrt{2}}$

Table No. 1.1 The equivalent lengths (L) for given end conditions [1]

Effective length of struts

The effective lengths according to IS: 800-1984 for various combinations of the end connections are

Effectively held in position at both ends and restrained against rotation at one end is given by,

$$0.80 L = 0.80 \times 4$$

The total height is approximately

$$13 \text{ FT} = 4 \text{ meters} = 0.80 \times 4 = \mathbf{3.2 \text{ Meters.}}$$

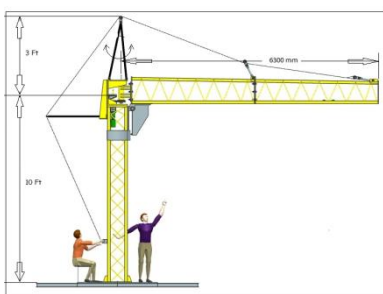


Fig. 1.2 Effective length

CHECKING THE BUCKLING OF BUILT-UP COLUMNS.

For built-up columns, we have to determine the safe load by Rankine's Formula having the column length of 3.2 meters.

Length, $l = 3.2$ meters

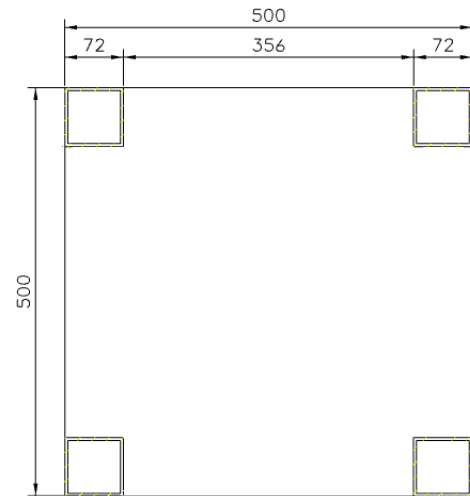


Fig 1.3 Buckling of built-up section

Factor of safety = 5

$$\text{Area, } a = 500 \times 500 = 2.5 \times 10^5 \text{ mm}^2$$

$$\text{Moment of Inertia, } I = 1.56 \times 10^{10}$$

Radius of gyration is given by,

$$= \sqrt{(1.56 \times 10^{10} / 2.5 \times 10^5)} = 249.7 \text{ mm Say} = 250 \text{ mm.}$$

For built-up columns, we have to determine the safe load by Rankine's Formula having the column length of 3.2 meters.

Length, $l = 3.2$ meters

Factor of safety = 5

$$\text{Area, } a = 500 \times 500 = 2.5 \times 10^5 \text{ mm}^2$$

$$\text{Moment of Inertia, } I = 1.56 \times 10^{10}$$

Radius of gyration is given by,

$$= \sqrt{(1.56 \times 10^{10} / 2.5 \times 10^5)} = 250 \text{ mm}$$

Therefore the Crippling load on the column is calculated by following,

$$P = (6c \times A) / (1 + a (L/K)^2)$$

[1] [6]

$$P = (320 \times 2.5 \times 10^3) / (1 + (1/7500) \times ((3.2 \times 10^3) / 250)) = 80 \times 10^3 \text{ KN}$$

$$\text{Safe load on the column} = (80 \times 10^3 / 5) = 16 \times 10^3 \text{ KN}$$

Thus column is **safer** for Buckling.

DESIGN OF COLUMN BASES:

Required width of gusset plate (B) = (Depth of girder + 2 x Cover plate thickness + 2 x Gusset plate thickness + Width of an angle connecting base plate + Clearance.

$$B = 500 + 2(16) + 2(16 + 100 + 14)$$

$$B = 792 \text{ mm and } A = 114 \text{ mm}$$

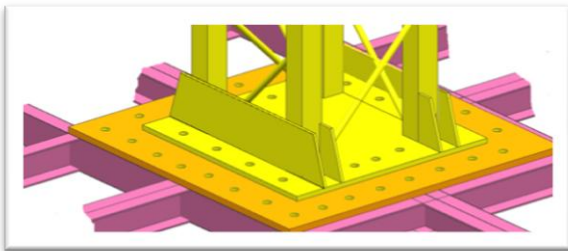


Fig 1.4 Column bases

Length of the gusset plate = $4.16 \times 10^3 / 792 = 5.25 \text{ mm}$ say 6mm

Area of a gusset plate provided = $6 \times 792 = 4752 \text{ mm}^2 > 4160 \text{ mm}^2$

Intensity of pressure between the plate and concrete = $25 \times 10^3 / 4.752 \times 10^3 = 5.26 \text{ MPa} < 6 \text{ MPa}$

The critical section for the bending moment is at the face of the angle.

Length of the critical section = $(115 - 16) + 14 = 113 \text{ mm}$

Considering unit width, Bending moment at critical section = $5.26 \times (113^2 / 2) = 33582.5 \text{ N mm}$

Plastic modulus required,

$$Z_p = M \times \gamma_{mo} / F_y = 33582.5 \times (1.1 / 250) = 147.8 \text{ Say } 148 \text{ mm}^3$$

Where,

$$\gamma_{mo} = \text{Partial safety factor} = 1.1$$

Elastic modulus,

$$Z_e = 148 / 1.14 = 129.8 \text{ say } 130 \text{ mm}^3$$

$$130 = 1 \times (t^2 / 6)$$

$$t = 27.9 \text{ mm say } 28 \text{ mm.}$$

Bending moment at critical section yy,

$$= W (B - 2a)^2 / 8 - W \times (a^2 / 2) = 5.26 (792 - (2 \times 114))^2 / 8 - 5.26 \times 114^2 / 2 = 174968.64 \text{ N.mm}$$

Plastic Modulus required; $Z_p = M \times \gamma_{mo} / F_y = 174968.64 \times 1.1 / 250 = 769.8 \text{ mm}^3 \text{ say } 770 \text{ mm}^3$

$$\text{Elastic Modulus; } Z_e = 770 / 1.14 = 675.43 = 1 \times t^2$$

$$t^2 = 174968.64 \times 6 = 63.66 \text{ mm say } 64 \text{ mm.}$$

The central portion has bending in two directions and also supported by web, hence thickness of plate is halved. Therefore, $t = (64 / 2) = 32 \text{ mm}$
So provided thickness of the base plate is 32 mm.

Design of Bolts: [1] [5]

The column section and the cover plates attached with the flanges as one unit. Therefore the bolts connecting gusset plate and column section are in single shear.

$$\sigma_p = P / A_{\text{total}} = P / (n \times A) = (25 \times 10^3) / (8 \times (\frac{\pi}{4}) \times (24^2)) = 7 \text{ N/mm}^2$$

Actual moment load applied on the plate and bolt is 65 KN
 $m = 65 \times 10^3 \text{ N m.}$

$\sigma_c =$ Number of bolts = 24

$\zeta =$ Average shear stress

$\zeta_{\text{max}} =$ Max shear stress

$$\zeta = (T \times r) / (A \times r^2)$$

$$\zeta = 1655.21 \text{ N/mm}^2$$

$$\zeta_{\text{max}} = (T \times r) / (A \times \epsilon \times r^2)$$

$$\zeta = 11605 \text{ N/mm}$$

Torsional load,

$$Pt = (T) / (r \times \epsilon \times r^2)$$

$$r = 300$$



Fig 1.5 Design of column bases with bolts

Shearing force = $(T \times r) / (\epsilon \times r^2) = 11 \text{ KN}$ and direct load = 25 KN.

$$\text{Resultant load} = \sqrt{(25^2 + 11^2)} = 28 \text{ KN}$$

$$\text{Maximum shear stress} = (28 \times 10^3) / 245 = 114.28 \text{ N/mm}^2$$

$$\text{Combined stress on the plate} = 122 \text{ N/mm}^2$$

Therefore, for M20 bolt size the tensile stress area = 245 mm²

For minimum shear strength of bolt grade 4.6 is 185 N/mm² which is 1.32 times **safe**.

1.2 Calculations of the maximum stresses induced in the members of the vertical structure of the crane.

Bending Stresses: By the simple theory of bending, $F_b = (M/I) \times y$

Allowable bending stress

As per IS: 800-1984, the allowable bending stress in tension F_{bt} is based upon the guaranteed minimum yield stress F_y of the steel. Considering factor of safety for bending stress is 1.5 times the bending stress.

Therefore,

$$F_{bt} = 0.66 \times F_y = 250 \text{ MPa}$$

$$\text{Means } F_{bt} = 165 \text{ MPa.}$$

And Shear stress

$F_a = V A_y / I_b$ and $F_s = 0.45 F_y$
 According to Standard table IS-800, data for Design of beams, coefficient of maximum deflection $K=1/3$.
 Therefore,
 $\Delta = (1/3) (W L^3 / EI)$
 $W = 10 \text{KN} = 10 \times 10^3 \text{ N}$, $L = 6300 \text{mm}$, $E = 200 \times 10^3 \text{ N/mm}^2$

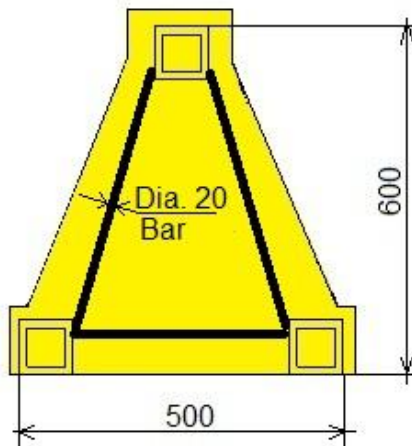


Fig 1.6 Moment of Inertia

$$I = 66.32 \text{ cm}^4 = 66.32 \text{ mm}^4$$

Load to be distributed considering 3 tubes,
 $10 \text{KN} / 3 = 3.33 \text{ KN}$ Say $3.5 \times 10^3 \text{ N}$

The maximum deflection at the end point will be

$$\Delta = (1/3) (W L^3 / EI) = 2199 \text{mm}$$

Calculating moment of Inertia:

$$I_A = B D^3 / 12$$

$$I_A = 8.5 \times 10^9 \text{ mm}^4 \text{ and } I_B = B D^3 / 12$$

$$I_B = 18 \times 10^9 \text{ mm}^4$$

Total Moment of Inertia,

$$I_C = I_A - 2I_B = 6.7 \times 10^9 \text{ say } 7 \times 10^9$$

Therefore the maximum deflection at the end is given by;

$$\Delta = (1/3) (W L^3 / EI) = (1/3) (10 \times 10^3 \times 6300^3) / (200 \times 10^3 \times 7 \times 10^9) = 0.622 \text{ mm}$$

The maximum deflection at the middle is given by;

$$\Delta = (1/3) (W L^3 / EI) = (1/3) (10 \times 10^3 \times 3150^3) / (200 \times 10^3 \times 7 \times 10^9) = 0.07 \text{ mm}$$

Considering the deflection practically, we can define every aspect separately and then add it to deflection formula,

The square section having an area = $8.54 \text{ cm}^2 = 854 \text{ mm}^2$

Considering horizontal section of $\phi 20 \text{ mm}$ bar having length 392 mm,

$$I_{\text{Bar}} = B D^3 / 12 = 20 \times (392)^3 / 12 = 100 \times 10^6 \text{ mm}^4$$

Therefore the total moment of inertia;

$$I_{\text{sq.section}} = 66.32 \text{ cm}^4 = 66.32 \times 10^4 \text{ mm}^4 \text{ and } I_{\text{Bar}} = 100 \times 10^6 \text{ mm}^4 = 201.98 \times 10^6 = 202 \times 10^6 \text{ mm}^4$$

The applied load = $10 \text{KN} = 10 \times 10^3 \text{ N}$ and $E = 200 \times 10^3 \text{ N/mm}^2$

The maximum deflection is given by;

$$\Delta = (1/3) (W L^3 / EI) = 20.63 \text{ mm say } 21 \text{ mm. Thus } \Delta / L = 21 / 6300 = 1/300$$

As per IS: 800, the allowable deflection should not exceed than $2/325$ of span for cantilever. Thus, **design** is safe for deflection.

1.3 Design considerations for drive section and calculations for the stresses and load carrying capacity of the main drive shaft of the crane. [1] [3] [5]

When a shaft is subjected to bending moment M_b and Torsional moment M_t , the bending stress σ_b and Torsional shear stress ζ given by,

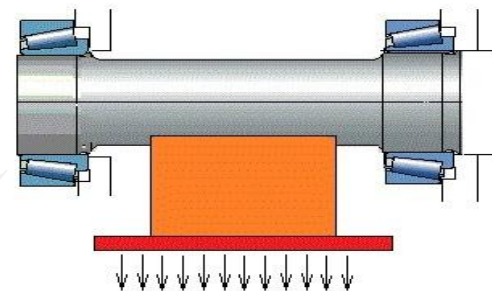


Fig 1.7 Design of drive shaft

M_t , the bending stress σ_b and Torsional shear stress ζ

$$\sigma_b = (M_b \times y) / I = 32 M_b / \pi d^3$$

$$\zeta = (M_t \times r) / y = 16 M_t / \pi d^3$$

Material - En8 with Yield strength of 620 N/mm^2 .

And $M_t = 63 \times 10^6 \text{ N mm}$

Shear stress can be taken as,

$$S_{st} = 0.577 \times 620 \text{ N/mm}^2 = 357.74 \text{ say } 358 \text{ N/mm}^2$$

Considering shaft diameter 110 mm.

$$\zeta = 16 M_t / \pi d^3 = (16 \times 63 \times 10^6) / (\pi \times 110^3) = 241.04 \text{ say } 241 \text{ N/mm}^2$$

$$\text{FOS} = \zeta / \zeta_{\text{max}} = 358 / 241 = 1.48 \text{ say } 1.5 \text{ times}$$

Considering bending strength,

$$R_B \times 660 = (15 \times 300) \times 330 = 2250 \text{ N and } R_A = (15 \times 300) - 2250 = 2250 \text{ N}$$

The shear force diagram above and values tabulated below.

$$F_A = +R_A = 2250 \text{ and } F_C = 2250 \text{ N and } F_D = 2250 - (15 \times 300) = -2250 \text{ N and } F_B = -2250 \text{ N}$$

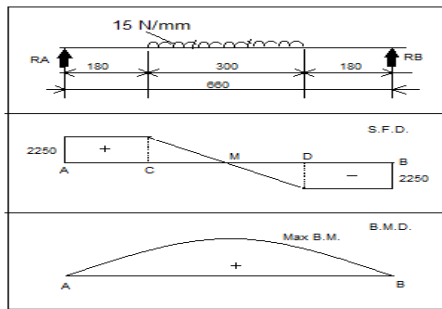


Fig. 1.8 SFD and BMD

On calculating $M_A = 0$, $M_C = 337500$, $M_D = -337500$

$$M_C = w l^2 / 8 \quad M_C = 168750 \text{ N-mm}$$

We have yield strength, $S_{yt} = 620 \text{ N/mm}^2$.

If we consider that the shaft is having keyway, then

$$\zeta_{\max} = (16 / \pi d^3) \times \sqrt{\{(K_b \times M_b)^2 + (K_t \times M_t)^2\}}$$

We know that,

$$S_{st} = 0.3 \times S_{yt} = 186 \text{ N/mm}^2 \quad \text{and} \quad S_{ut} = 0.18 \times S_{yt} = 111.6 \text{ N/mm}^2$$

According to standard, the following constants taken in to consideration.

$$K_b = 1.5 \quad \text{and} \quad K_t = 1$$

$$\zeta_{\max} = (16 / \pi d^3) \times \sqrt{\{(K_b \times M_b)^2 + (K_t \times M_t)^2\}} = 241.06 \text{ say } 241 \text{ N/mm}^2$$

Factor of safety = $241 / 114 = 2.11$ means design for shaft in tension and shear is safe.

1.4 Procedure for selection of Wire rope for safe working of crane. [1] [5]

Assuming number of wire ropes 'n'. The force acting on each wire rope compares weight of the material to be raised, the weight of the wire rope and the force due to the acceleration of the material and the wire rope.

$$\text{The weight of the material raised by each wire rope, } (10 \times 1000/n) \text{ N} \quad (1)$$

$$\text{From table mass of 100 m wire rope is 34.6 Kg, since the weight is,} \\ = 34.6 \times 9.81 \times 40 / 100 = 135.77 \text{ N} \quad (2)$$

The mass of the material raised by each wire rope is $(10000 / (n \times 9.81))$ and that of each wire rope is, $(34.6 \times 40) / 100$. Therefore, due to each acceleration is,

$$= [(10000/(n \times 9.81)) + (34.6 \times 40)/(100)] \times 1 = [(1019.36/n) + 13.84] \text{ N} \quad (3)$$

From below table, the breaking strength of the wire rope is 44 KN.

Assuming the FOS to be 10.

$$(44000/10) = (10000 / n) + 135.77 + (1019.36 / n) + 13.84 = 2.59 \text{ say } 3$$

Nominal Diameter (mm)	Approximate Mass (kg/100 m)	Minimum breaking load to tensile designation of wires of (KN)		
		1230	1420	1570
6	12.5	13.6	15.7	17.4
7	17.0	18.5	21	24
8	22.1	24	28	31
9	28.0	31	35	39
10	34.6	38	44	48
11	41.9	46	53	58
12	49.8	54	63	69

Table 1.2 Breaking load and mass for 6x9 (12/6/1) Construction wire ropes as per IS

So two wire ropes of $\phi 10$ mm taken in to account. On an availability of material or wire rope selected is $\frac{1}{2}$ " or $\phi 12$ mm. considering, the diameter is selected is much safer side. Considering the maximum tension in the wire rope is 30 KN.

The mechanical properties of wire ropes considering IS-800: 2266-1981, Steel wire ropes general engineering purpose specifications, the nominal diameter (dr) of the wire rope indicate the diameter of the smallest circle enclosing the wire rope. The designation of the wires, such as 1570 or 1770 indicates the minimum ultimate tensile strength (in N/mm^2) of the individual wires used for making of the rope.

According the table for 6x19 (12/6x1) construction wire ropes for nominal diameter 81 KN for 1570 N/mm^2 . This proves much safer for working.

1.5 Procedure to find out knuckle pin joint for wire rope. [5]

The total load carrying at the end of the mast is subjected to maximum stresses induced at the joining part of the rope and horizontal mast. This load is carried by means of the knuckle pin.

Considering the factor of safety to total load of 5 KN.



Fig 1.9 Design of Knuckle pin

The torque (T) transmitted by pulley,

$$T = 30 \times 5 = 1.5 \times 10^5 \text{ N mm}$$

D = Minimum shaft diameter

$$\text{Torque, } T = 1.5 \times 10^5 \text{ N mm}$$

$$\zeta = 45 \text{ MPa}$$

D = to be find out

$$T = (\pi/16) \times \zeta \times D^3$$

$$D^3 = (1.5 \times 10^5 \times 16) / (\pi \times 45) = 26 \text{ mm.}$$

On an availability of the standard Knuckle pin at market, we have bought 30 mm knuckle pin on safer side.

1.6 FEA analysis of Crane with the help of NX 8 NASTRAN

Simple mathematical model can be solved analytically, but more complex model requires use of numerical methods. FEA is one of the numerical methods used to solve complex mathematical problem. The entire solution domain must be discretized into simply shaped sub domain called as elements. NX 8 NASTRAN software is used for the analysis of the Material Handling Crane, which is based on the FEA method.

STEPS IN FINITE ELEMENT ANALYSIS

1. 3D modeling of Material Handling Crane.

The parts have been created with parametric modeling in 3D using NX-8 software

The created 3D model is saved in part.prt file format, as this file format is suitable during importing this model for meshing in NX NASTRAN software.

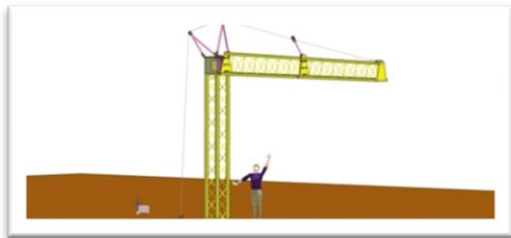


Fig 1.10 3D model

2. Meshing of the 3D model of Drive section.

In simple term meshing means connecting elements with each other. Elements are the building blocks of the finite element analysis. Meshing is carried out by using NX NASTRAN software which is largely used for



Fig 1.11 Meshed model

meshing. Model is meshed by using SOLID 45 element and with 10 element size.

Total 41231 nodes and 410232 elements were created after meshing.

3. *Material properties:* After completion of meshing material properties are assigned to meshed model.

These properties are listed below.

Material used-Steel En 8

Young's Modulus- $2.1 \times 10^5 \text{ N/mm}^2$

Poisson's Ratio -0.26

4. *Applying constraints:* After modeling of the parts the proper constraining of the part is carried out.

5. *Structural loading on the component:* Structural loading means applying proper stresses on the drive section parts.

6. *Solving of the FEA model:* In this case Post Solver of NX NASTRAN solves the matrices internally. In this case, the solver solves mathematical model and gives the required results

7. Results and its physical interpretation

After applying material properties and boundary conditions, problem was solved by the NASTRAN solver. NASTRAN solver formulates the governing structural stress strain equations for each and every element those formulated governing equation are solved for deformation.

SUMMARIZED PRINCIPAL STRESS DATA USING FEA:

The following images shows the various results for the vertical section, horizontal boom, drive section box, drive shaft, pulley bracket, etc.



Fig 1.12 Deflection of horizontal boom

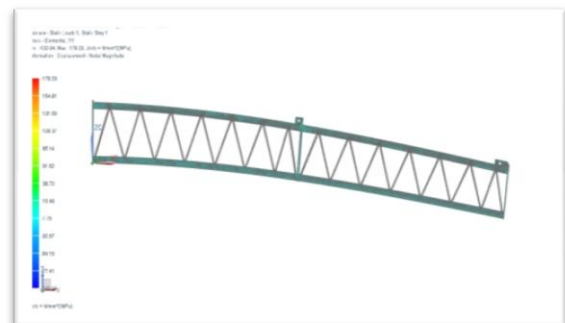


Fig 1.13 Von mises stress at horizontal boom

Chapter 2: Manufacturing

According to the design considerations the manufacturing of the Material Handling Crane was carried out at “Maharatna Steel Industries, Sakharale”. Before the manufacturing of the crane, following things considered;

- | | |
|------------------------------|-------------------------|
| 1) Availability of the space | 2) Working machineries. |
| 3) Manufacturing knowledge | 4) Handling equipment’s |
| 5) Their clients. | 6) Manpower |
| 7) Technical knowledge | 8) Transportation |

MANUFACTURING PROCESS

Manufacturing of a crane was divided in to following parts;

1. Fabrication of vertical structure
2. Fabrication of horizontal boom structure
3. Fabrication and manufacturing of drive section.
4. Fabrication of supplementary parts like temporary base, rope support bracket, support plate.

The following photos elaborate the fabrication of a steel structure, manufacturing of the shaft, bearing plates, drive box setion.

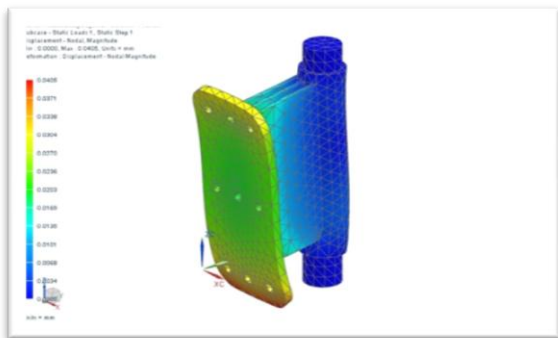


Fig 1.14 Displacement at drive shaft

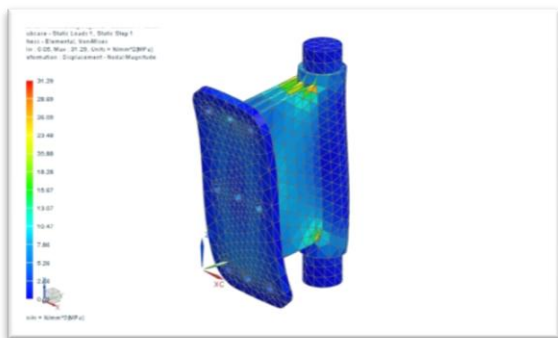


Fig 1.15 Von-mises stresses at drive shaft

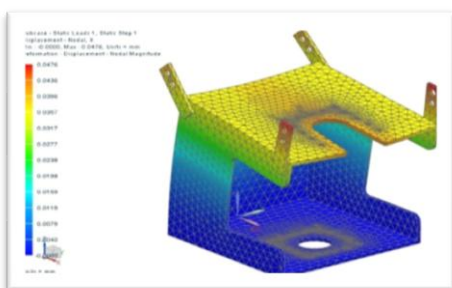


Fig 1.16 Displacement at drive Box

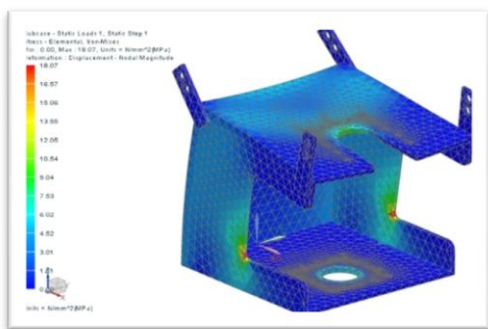


Fig 1.17 Von-mises stresses at drive Box



Fig. 2.1 Manufacturing of Drive Box



Fig. 2.3 Manufacturing of Vertical Section



Fig 2.4 manufacturing of Horizontal Section

CHAPTER 3: CONCLUSION

In this dissertation work an attempt has been made for **Design and Development of Material Handling Crane**. Various design considerations made by previous design parameters and analyzed these models for better results.

The advent of innovative crane design has brought with the provision and tendency for being used as items of lifting and material handling equipment. This situation will present a whole new set of health and safety hazards to crane owners, operators.

. Following mentioned are some of the conclusion made on the Design and Development of Material Handling crane

- 1 Designing of this new concept i.e. a crane balancing using a wire rope or without counter weight.
- 2 Good ergonomic design theme.
- 3 The overall working space is more in this type of design.
- 4 Manufacturing or fabrication is much easier and suitable for working.
- 5 Comparatively assembly time is less considering the overall assembly.
- 6 Working capability as per initial requirement completed.
- 7 User friendly working.
- 8 Less dismantling time.
- 9 Low manufacturing cost.
- 10 Nice protective painting yellow coat.

CHAPTER 4: REFERENCES

4.1 Books:

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Chapter 3: Assembly and Testing

3.1 Assembly of the crane structure

The total assembly of the crane structure is carried out after manufacturing and fabrication of all components. The heavy structure is assembled at free space at Maharatna Steel industries. Following procedure is carried out during the assembly of the Crane structure.



Fig. 3.2 Complete assembly of crane

3.2 Inspection and Testing

After assembly of the crane structure the testing procedure is carried out. The inspection report is generated with reference to IS: 807 and IS: 3177 for cranes. For testing purpose, the small weights weighing 20 Kg each has been taken. Load is gradually increased while testing of the crane.



Fig. 3.3 Testing of Crane